Active Damping of Rotating Platforms using Integral Force Feedback - Matlab Computation

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February 20, 2021

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This document gathers the Matlab code used to for the conference paper [1] and the journal paper [3].

It is structured in several sections:

- Section 1: presents a simple model of a rotating suspended platform that will be used throughout this study.
- Section 2: explains how the unconditional stability of IFF is lost due to Gyroscopic effects induced by the rotation.
- Section 3: suggests a simple modification of the control law such that damping can be added to the suspension modes in a robust way.
- Section 4: proposes to add springs in parallel with the force sensors to regain the unconditional stability of IFF.
- Section 5: compares both proposed modifications to the classical IFF in terms of damping authority and closed-loop system behavior.
- Section 6: contains the notations used for both the Matlab code and the paper

The matlab code is accessible on Zonodo and Github [2]. It can also be download as a .zip file here.

To run the Matlab code, go in the **matlab** directory and run the following Matlab files corresponding to each section.

Table 0.1:	Paper'	s sections	and	corresponding	Matlab	files
------------	--------	------------	-----	---------------	--------	-------

Sections	Matlab File
Section 1	<pre>s1_system_description.m</pre>
Section 2	<pre>s2_iff_pure_int.m</pre>
Section 3	s3_iff_hpf.m
Section 4	s4_iff_kp.m
Section 5	s5_act_damp_comparison.m

1 System Description and Analysis

1.1 System description

The system consists of one 2 degree of freedom translation stage on top of a spindle (figure 1.1).



Figure 1.1: Schematic of the studied system

The control inputs are the forces applied by the actuators of the translation stage (F_u and F_v). As the translation stage is rotating around the Z axis due to the spindle, the forces are applied along \vec{i}_u and \vec{i}_v .

1.2 Equations

Based on the Figure 1.1, the equations of motions are:

Important

$$\begin{bmatrix} d_u \\ d_v \end{bmatrix} = G_d \begin{bmatrix} F_u \\ F_v \end{bmatrix}$$
(1.1)

Where G_d is a 2×2 transfer function matrix.

$$G_d = \frac{1}{k} \frac{1}{G_{dp}} \begin{bmatrix} G_{dz} & G_{dc} \\ -G_{dc} & G_{dz} \end{bmatrix}$$
(1.2)

With:

$$G_{dp} = \left(\frac{s^2}{\omega_0^2} + 2\xi \frac{s}{\omega_0} + 1 - \frac{\Omega^2}{\omega_0^2}\right)^2 + \left(2\frac{\Omega}{\omega_0}\frac{s}{\omega_0}\right)^2$$
(1.3)

$$G_{dz} = \frac{s^2}{\omega_0^2} + 2\xi \frac{s}{\omega_0} + 1 - \frac{\Omega^2}{\omega_0^2}$$
(1.4)

$$G_{dc} = 2\frac{\Omega}{\omega_0} \frac{s}{\omega_0} \tag{1.5}$$

1.3 Numerical Values

Let's define initial values for the model.

```
        K = 1;
        % Actuator Stiffness [N/m]
        Matlab

        c = 0.05; % Actuator Damping [N/(m/s)]
        m = 1;
        % Payload mass [kg]
```

```
xi = c/(2*sqrt(k*m));
w0 = sqrt(k/m); % [rad/s]
```

1.4 Campbell Diagram

The Campbell Diagram displays the evolution of the real and imaginary parts of the system as a function of the rotating speed.

Matlab

It is shown in Figures 1.2 and 1.3, and one can see that the system becomes unstable for $\Omega > \omega_0$ (the real part of one of the poles becomes positive).

1.5 Simscape Model

In order to validate all the equations of motion, a Simscape model of the same system has been developed. The dynamics of the system can be identified from the Simscape model and compare with the analytical model.



Figure 1.2: Campbell Diagram - Real Part



Figure 1.3: Campbell Diagram - Imaginary Part

The rotating speed for the Simscape Model is defined.

W = 0.1; % Rotation Speed [rad/s]

open('rotating_frame.slx');

__ Matlab _

The transfer function from $[F_u, F_v]$ to $[d_u, d_v]$ is identified from the Simscape model.

Matlab Matlab mdl = 'rotating_frame'; %% Input/Output definition clear io; io_i = 1; io(io_i) = linio([mdl, '/K'], 1, 'openinput'); io_i = io_i + 1; io(io_i) = linio([mdl, '/G'], 2, 'openoutput'); io_i = io_i + 1;

Matlab _

G = linearize(mdl, io, 0);
%% Input/Output definition
G.InputName = {'Fu', 'Fv'};
G.OutputName = {'du', 'dv'};

The same transfer function from $[F_u, F_v]$ to $[d_u, d_v]$ is written down from the analytical model.

```
 \begin{array}{c} \mbox{Matlab} \\ \mbox{Gth} = (1/k)/(((s^2)/(w0^2) + 2*xi*s/w0 + 1 - (W^2)/(w0^2))^2 + (2*W*s/(w0^2))^2) * \dots \\ [(s^2)/(w0^2) + 2*xi*s/w0 + 1 - (W^2)/(w0^2), 2*W*s/(w0^2) ; \dots \\ -2*W*s/(w0^2), (s^2)/(w0^2) + 2*xi*s/w0 + 1 - (W^2)/(w0^2)]; \end{array}
```

Both transfer functions are compared in Figure 1.4 and are found to perfectly match.

1.6 Effect of the rotation speed

The transfer functions from $[F_u, F_v]$ to $[d_u, d_v]$ are identified for the following rotating speeds.

```
________Matlab ______
Ws = [0, 0.2, 0.7, 1.1]*w0; % Rotating Speeds [rad/s]
```

Matlab		
Gs = {zeros(2, 2, length(Ws))};		
<pre>for W_i = 1:length(Ws) W = Ws(W_i);</pre>		
Gs(:, :, W_i) = {(1/k)/(((s ²)/(w0 ²) + 2*xi*s/w0 + 1 - (W ²)/(w0 ²)) ² + (2*W*s/(w0 ²)) ²) * [(s ²)/(w0 ²) + 2*xi*s/w0 + 1 - (W ²)/(w0 ²), 2*W*s/(w0 ²) ; -2*W*s/(w0 ²), (s ²)/(w0 ²) + 2*xi*s/w0 + 1 - (W ²)/(w0 ²)];		
end		



Figure 1.4: Bode plot of the transfer function from $[F_u, F_v]$ to $[d_u, d_v]$ as identified from the Simscape model and from an analytical model

They are compared in Figures 1.5 and 1.6.



Figure 1.5: Comparison of the transfer functions from $[F_u, F_v]$ to $[d_u, d_v]$ for several rotating speed -Direct Terms



Figure 1.6: Comparison of the transfer functions from $[F_u, F_v]$ to $[d_u, d_v]$ for several rotating speed - Coupling Terms

2 Problem with pure Integral Force Feedback

Force sensors are added in series with the two actuators (Figure 2.1).

Two identical controllers K_F are used to feedback each of the sensed force to its associated actuator.



Figure 2.1: System with added Force Sensor in series with the actuators

2.1 Plant Parameters

Let's define initial values for the model.

```
k = 1; % Actuator Stiffness [N/m]
c = 0.05; % Actuator Damping [N/(m/s)]
m = 1; % Payload mass [kg]
```

```
xi = c/(2*sqrt(k*m));
w0 = sqrt(k/m); % [rad/s]
```

Matlab

Matlab .

2.2 Equations

The sensed forces are equal to:

$$\begin{bmatrix} f_u \\ f_v \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} F_u \\ F_v \end{bmatrix} - (cs+k) \begin{bmatrix} d_u \\ d_v \end{bmatrix}$$
(2.1)

Which then gives:

Important

$$\begin{bmatrix} f_u \\ f_v \end{bmatrix} = G_f \begin{bmatrix} F_u \\ F_v \end{bmatrix}$$
(2.2)

$$G_{fp} = \left(\frac{s^2}{\omega_0^2} + 2\xi \frac{s}{\omega_0} + 1 - \frac{\Omega^2}{\omega_0^2}\right)^2 + \left(2\frac{\Omega}{\omega_0}\frac{s}{\omega_0}\right)^2$$
(2.4)

$$G_{fz} = \left(\frac{s^2}{\omega_0^2} - \frac{\Omega^2}{\omega_0^2}\right) \left(\frac{s^2}{\omega_0^2} + 2\xi \frac{s}{\omega_0} + 1 - \frac{\Omega^2}{\omega_0^2}\right) + \left(2\frac{\Omega}{\omega_0}\frac{s}{\omega_0}\right)^2 \tag{2.5}$$

$$G_{fc} = \left(2\xi \frac{s}{\omega_0} + 1\right) \left(2\frac{\Omega}{\omega_0} \frac{s}{\omega_0}\right) \tag{2.6}$$

2.3 Comparison of the Analytical Model and the Simscape Model

The rotation speed is set to $\Omega = 0.1\omega_0$.

And the transfer function from $[F_u, F_v]$ to $[f_u, f_v]$ is identified using the Simscape model.

Matlab mdl = 'rotating_frame'; %% Input/Output definition clear io; io_i = 1; io(io_i) = linio([mdl, '/K'], 1, 'openinput'); io_i = io_i + 1; io(io_i) = linio([mdl, '/G'], 1, 'openoutput'); io_i = io_i + 1;

	Matlab
<pre>Giff = linearize(mdl, io, 0);</pre>	
<pre>%% Input/Output definition Giff.InputName = {'Fu', 'Fv'}; Giff.OutputName = {'fu', 'fv'};</pre>	

The same transfer function from $[F_u, F_v]$ to $[f_u, f_v]$ is written down from the analytical model.

Matlad
$Giff_th = 1/(((s^2)/(w0^2) + 2*xi*s/w0 + 1 - (W^2)/(w0^2))^2 + (2*W*s/(w0^2))^2) * \dots$
[(s^2/w0^2 - W^2/w0^2)*((s^2)/(w0^2) + 2*xi*s/w0 + 1 - (W^2)/(w0^2)) + (2*W*s/(w0^2))^2, - (2*xi*s/w0 +
↔ 1)*2*W*s/(w0 ²) ;
(2*xi*s/w0 + 1)*2*W*s/(w0^2), (s^2/w0^2 - W^2/w0^2)*((s^2)/(w0^2) + 2*xi*s/w0 + 1 - (W^2)/(w0^2))+ (2*W*s/(w0^2))^2];

The two are compared in Figure 2.2 and found to perfectly match.



Figure 2.2: Comparison of the transfer functions from $[F_u, F_v]$ to $[f_u, f_v]$ between the Simscape model and the analytical one

2.4 Effect of the rotation speed

The transfer functions from $[F_u, F_v]$ to $[f_u, f_v]$ are identified for the following rotating speeds.

Ws = [0, 0.2, 0.7]*w0; % Rotating Speeds [rad/s]	
	Matlab
<pre>Gsiff = {zeros(2, 2, length(Ws))};</pre>	
<pre>for W_i = 1:length(Ws) W = Ws(W_i);</pre>	
Gsiff(:, :, W_i) = {1/(((s^2)/(w0^2) + 2*xi*s/w0 + 1 - [(s^2/w0^2 - W^2/w0^2)*((s^2)/(w0^2)) → 1)*2*W*s/(w0^2); (2*xi*s/w0 + 1)*2*W*s/(w0^2), (s^2/w → (2*W*s/(w0^2))^2];	(W ²)/(W0 ²)) ² + (2*W*s/(W0 ²)) ²) * + 2*xi*s/w0 + 1 - (W ²)/(W0 ²)) + (2*W*s/(W0 ²)) ² , - (2*xi*s/w0 + 0 ² - W ² /w0 ²)*((s ²)/(W0 ²) + 2*xi*s/w0 + 1 - (W ²)/(W0 ²))+
end	

Matlah

The obtained transfer functions are shown in Figure 2.3.



Figure 2.3: Comparison of the transfer functions from $[F_u, F_v]$ to $[f_u, f_v]$ for several rotating speed

2.5 Decentralized Integral Force Feedback

The decentralized IFF controller consists of pure integrators:

$$K_{\rm IFF}(s) = \frac{g}{s} \begin{bmatrix} 1 & 0\\ 0 & 1 \end{bmatrix}$$

$$\tag{2.7}$$

The Root Locus (evolution of the poles of the closed loop system in the complex plane as a function of g) is shown in Figure 2.4. It is shown that for non-null rotating speed, one pole is bound to the right-half plane, and thus the closed loop system is unstable.



Figure 2.4: Root Locus for the Decentralized Integral Force Feedback controller. Several rotating speed are shown.

3 Integral Force Feedback with an High Pass Filter

. Matlab _

_ Matlab _

3.1 Plant Parameters

Let's define initial values for the model.

```
k = 1; % Actuator Stiffness [N/m]
c = 0.05; % Actuator Damping [N/(m/s)]
m = 1; % Payload mass [kg]
```

```
xi = c/(2*sqrt(k*m));
w0 = sqrt(k/m); % [rad/s]
```

3.2 Modified Integral Force Feedback Controller

Let's modify the initial Integral Force Feedback Controller ; instead of using pure integrators, pseudo integrators (i.e. low pass filters) are used:

_ Matlab _

$$K_{\rm IFF}(s) = g \frac{1}{\omega_i + s} \begin{bmatrix} 1 & 0\\ 0 & 1 \end{bmatrix}$$
(3.1)

where ω_i characterize down to which frequency the signal is integrated.

Let's arbitrary choose the following control parameters:

$$g = 2;$$

And the following rotating speed.

The obtained Loop Gain is shown in Figure 3.1.



Figure 3.1: Loop Gain for the modified IFF controller

3.3 Root Locus

As shown in the Root Locus plot (Figure 3.2), for some value of the gain, the system remains stable.



Figure 3.2: Root Locus for the modified IFF controller

3.4 What is the optimal ω_i and g?

In order to visualize the effect of ω_i on the attainable damping, the Root Locus is displayed in Figure 3.4 for the following ω_i :

Matlab

wis = [0.01, 0.1, 0.5, 1]*w0; % [rad/s]



Figure 3.3: Root Locus for the modified IFF controller - Zoom



Figure 3.4: Root Locus for the modified IFF controller (zoomed plot on the left)



Figure 3.5: Root Locus for the modified IFF controller (zoomed plot on the left)

For the controller

$$K_{\rm IFF}(s) = g \frac{1}{\omega_i + s} \begin{bmatrix} 1 & 0\\ 0 & 1 \end{bmatrix}$$
(3.2)

The gain at which the system becomes unstable is

$$g_{\max} = \omega_i \left(\frac{\omega_0^2}{\Omega^2} - 1\right) \tag{3.3}$$

While it seems that small ω_i do allow more damping to be added to the system (Figure 3.4), the control gains may be limited to small values due to (3.3) thus reducing the attainable damping.

There must be an optimum for ω_i . To find the optimum, the gain that maximize the simultaneous damping of the mode is identified for a wide range of ω_i (Figure 3.6).

Matlab _

```
wis = logspace(-2, 1, 100)*w0; % [rad/s]
opt_xi = zeros(1, length(wis)); % Optimal simultaneous damping
opt_gain = zeros(1, length(wis)); % Corresponding optimal gain
for wi_i = 1:length(wis)
    wi = wis(wi_i);
    Kiff = 1/(s + wi)*eye(2);
    fun = @(g)computeSimultaneousDamping(g, Giff, Kiff);
    [g_opt, xi_opt] = fminsearch(fun, 0.5*wi*((w0/W)^2 - 1));
    opt_xi(wi_i) = 1/xi_opt;
    opt_gain(wi_i) = g_opt;
end
```



Figure 3.6: Simultaneous attainable damping of the closed loop poles as a function of ω_i

4 IFF with a stiffness in parallel with the force sensor

4.1 Schematic

In this section additional springs in parallel with the force sensors are added to counteract the negative stiffness induced by the rotation.



Figure 4.1: Studied system with additional springs in parallel with the actuators and force sensors

In order to keep the overall stiffness $k = k_a + k_p$ constant, a scalar parameter α ($0 \le \alpha < 1$) is defined to describe the fraction of the total stiffness in parallel with the actuator and force sensor

$$k_p = \alpha k, \quad k_a = (1 - \alpha)k \tag{4.1}$$

4.2 Equations

Important

$$\begin{bmatrix} f_u \\ f_v \end{bmatrix} = G_k \begin{bmatrix} F_u \\ F_v \end{bmatrix}$$
(4.2)

$$\begin{bmatrix} f_u \\ f_v \end{bmatrix} = \frac{1}{G_{kp}} \begin{bmatrix} G_{kz} & -G_{kc} \\ G_{kc} & G_{kz} \end{bmatrix} \begin{bmatrix} F_u \\ F_v \end{bmatrix}$$
(4.3)

With:

$$G_{kp} = \left(\frac{s^2}{\omega_0^2} + 2\xi \frac{s}{\omega_0^2} + 1 - \frac{\Omega^2}{\omega_0^2}\right)^2 + \left(2\frac{\Omega}{\omega_0}\frac{s}{\omega_0}\right)^2$$
(4.4)

$$G_{kz} = \left(\frac{s^2}{\omega_0^2} - \frac{\Omega^2}{\omega_0^2} + \alpha\right) \left(\frac{s^2}{\omega_0^2} + 2\xi \frac{s}{\omega_0^2} + 1 - \frac{\Omega^2}{\omega_0^2}\right) + \left(2\frac{\Omega}{\omega_0}\frac{s}{\omega_0}\right)^2$$
(4.5)

$$G_{kc} = \left(2\xi \frac{s}{\omega_0} + 1 - \alpha\right) \left(2\frac{\Omega}{\omega_0} \frac{s}{\omega_0}\right) \tag{4.6}$$

If we compare G_{kz} and G_{fz} , we see that the spring in parallel adds a term α . In order to have two complex conjugate zeros (instead of real zeros):

$$\alpha > \frac{\Omega^2}{\omega_0^2} \quad \Leftrightarrow \quad k_p > m\Omega^2 \tag{4.7}$$

4.3 Plant Parameters

Let's define initial values for the model.

```
      k = 1;
      % Actuator Stiffness [N/m]

      c = 0.05; % Actuator Damping [N/(m/s)]

      m = 1;
      % Payload mass [kg]
```

_ Matlab _

xi = c/(2*sqrt(k*m)); w0 = sqrt(k/m); % [rad/s]

4.4 Comparison of the Analytical Model and the Simscape Model

The same transfer function from $[F_u, F_v]$ to $[f_u, f_v]$ is written down from the analytical model.

open('rotating_frame.slx');

Matlab _

4.5 Effect of the parallel stiffness on the IFF plant

The rotation speed is set to $\Omega = 0.1\omega_0$.

```
W = 0.1*w0; % [rad/s]
```

Giff_th.InputName = {'Fu', 'Fv'}; Giff_th.OutputName = {'fu', 'fv'};

Matlab _

And the IFF plant (transfer function from $[F_u, F_v]$ to $[f_u, f_v]$) is identified in three different cases:

- without parallel stiffness
- with a small parallel stiffness $k_p < m\Omega^2$
- with a large parallel stiffness $k_p > m\Omega^2$

The results are shown in Figure 4.3.

One can see that for $k_p > m\Omega^2$, the systems shows alternating complex conjugate poles and zeros.

_ Matlab _

```
kp = 0;
```

w0p = sqrt((k + kp)/m); xip = c/(2*sqrt((k+kp)*m));



Figure 4.2: Comparison of the transfer functions from $[F_u, F_v]$ to $[f_u, f_v]$ between the Simscape model and the analytical one

```
Giff = 1/( (s^2/w0p^2 + 2*xip*s/w0p + 1 - W^2/w0p^2)^2 + (2*(s/w0p)*(W/w0p))^2 ) * [ ...
(s^2/w0p^2 + kp/(k + kp) - W^2/w0p^2)*(s^2/w0p^2 + 2*xip*s/w0p + 1 - W^2/w0p^2) + (2*(s/w0p)*(W/w0p))^2, -(2*xip*s/w0p +

→ k/(k + kp))*(2*(s/w0p)*(W/w0p));
(2*xip*s/w0p + k/(k + kp))*(2*(s/w0p)*(W/w0p)), (s^2/w0p^2 + kp/(k + kp) - W^2/w0p^2)*(s^2/w0p^2 + 2*xip*s/w0p + 1 -

→ W^2/w0p^2) + (2*(s/w0p)*(W/w0p))^2];
```

 $kp = 0.5 + m + W^{2};$ k = 1 - kp; _ Matlab

w0p = sqrt((k + kp)/m); xip = c/(2*sqrt((k+kp)*m));

Giff_s = 1/((s²/w0p² + 2*xip*s/w0p + 1 - W²/w0p²)² + (2*(s/w0p)*(W/w0p))²) * [... (s²/w0p² + kp/(k + kp) - W²/w0p²)*(s²/w0p² + 2*xip*s/w0p + 1 - W²/w0p²) + (2*(s/w0p)*(W/w0p))², -(2*xip*s/w0p + → k/(k + kp))*(2*(s/w0p)*(W/w0p)); (2*xip*s/w0p + k/(k + kp))*(2*(s/w0p)*(W/w0p)), (s²/w0p² + kp/(k + kp) - W²/w0p²)*(s²/w0p² + 2*xip*s/w0p + 1 -→ W²/w0p²) + (2*(s/w0p)*(W/w0p))²];





Figure 4.3: Transfer function from $[F_u, F_v]$ to $[f_u, f_v]$ for $k_p = 0$, $k_p < m\Omega^2$ and $k_p > m\Omega^2$

4.6 IFF when adding a spring in parallel

In Figure 4.4 is displayed the Root Locus in the three considered cases with

$$K_{\rm IFF} = \frac{g}{s} \begin{bmatrix} 1 & 0\\ 0 & 1 \end{bmatrix} \tag{4.8}$$

One can see that for $k_p > m\Omega^2$, the root locus stays in the left half of the complex plane and thus the control system is unconditionally stable.

Thus, decentralized IFF controller with pure integrators can be used if:

$$k_p > m\Omega^2 \tag{4.9}$$



Figure 4.5: Root Locus

4.7 Effect of k_p on the attainable damping

However, having large values of k_p may decrease the attainable damping.

To study the second point, Root Locus plots for the following values of k_p are shown in Figure 4.6.

Matlab

kps = [2, 20, 40]*m*W^2;

It is shown that large values of k_p decreases the attainable damping.



Figure 4.6: Root Locus plot

```
Matlab

alphas = logspace(-2, 0, 100);

opt_xi = zeros(1, length(alphas)); % Optimal simultaneous damping

opt_gain = zeros(1, length(alphas)); % Corresponding optimal gain

Kiff = 1/s*eye(2);

for alpha_i = 1:length(alphas)

kp = alphas(alpha_i);

k = 1 - alphas(alpha_i);

w0p = sqrt((k + kp)/m);

xip = c/(2*sqrt((k+kp)*m));

Giff = 1/( (s^2/w0p^2 + 2*xip*s/w0p + 1 - W^2/w0p^2)^2 + (2*(s/w0p)*(W/w0p))^2) * [ ...

    (s^2/w0p^2 + kp/(k + kp) - W^2/w0p^2)(s^2/w0p^2 + 2*xip*s/w0p + 1 - W^2/w0p^2) + (2*(s/w0p)*(W/w0p))^2, -(2*xip*s/w0p

    → + k/(k + kp))*(2*(s/w0p)*(W/w0p));

    (2*xip*s/w0p + k/(k + kp))*(2*(s/w0p)*(W/w0p)), (s^2/w0p^2 + kp/(k + kp) - W^2/w0p^2)*(s^2/w0p^2 + 2*xip*s/w0p + 1 -

    → W^2/w0p^2) + (2*(s/w0p)*(W/w0p))*2];

fun = 0(g)computeSimultaneousDamping(g, Giff, Kiff);

    [g_opt, xi_opt] = fminsearch(fun, 2);

    opt_xi(alpha_i) = 1/xi_opt;

    opt_gain(alpha_i) = g_opt;

end
```



Figure 4.7: Attainable damping ratio and corresponding controller gain for different parameter α

5 Comparison

Two modifications to adapt the IFF control strategy to rotating platforms have been proposed. These two methods are now compared in terms of added damping, closed-loop compliance and transmissibility.

_ Matlab _

_ Matlab _

5.1 Plant Parameters

Let's define initial values for the model.

```
k = 1; % Actuator Stiffness [N/m]
c = 0.05; % Actuator Damping [N/(m/s)]
m = 1; % Payload mass [kg]
```

```
xi = c/(2*sqrt(k*m));
w0 = sqrt(k/m); % [rad/s]
```

The rotating speed is set to $\Omega = 0.1\omega_0$.

W = 0.1 * w0;

_ Matlab _

5.2 Root Locus

IFF with High Pass Filter

```
wi = 0.1*w0; % [rad/s]
Giff = 1/(((s<sup>2</sup>)/(w0<sup>2</sup>) + 2*xi*s/w0 + 1 - (W<sup>2</sup>)/(w0<sup>2</sup>))<sup>2</sup> + (2*W*s/(w0<sup>2</sup>))<sup>2</sup>) * ...
[(s<sup>2</sup>/w0<sup>2</sup> - W<sup>2</sup>/w0<sup>2</sup>)*((s<sup>2</sup>)/(w0<sup>2</sup>) + 2*xi*s/w0 + 1 - (W<sup>2</sup>)/(w0<sup>2</sup>)) + (2*W*s/(w0<sup>2</sup>))<sup>2</sup>, - (2*xi*s/w0 + 1)*2*W*s/(w0<sup>2</sup>))
→ ; ...
(2*xi*s/w0 + 1)*2*W*s/(w0<sup>2</sup>), (s<sup>2</sup>/w0<sup>2</sup> - W<sup>2</sup>/w0<sup>2</sup>)*((s<sup>2</sup>)/(w0<sup>2</sup>) + 2*xi*s/w0 + 1 - (W<sup>2</sup>)/(w0<sup>2</sup>))+ (2*W*s/(w0<sup>2</sup>))<sup>2</sup>];
```

_ Matlab __

IFF With parallel Stiffness





Figure 5.1: Root Locus plot - Comparison of IFF with additional high pass filter, IFF with additional parallel stiffness

5.3 Controllers - Optimal Gains

In order to compare to three considered Active Damping techniques, gains that yield maximum damping of all the modes are computed for each case.

The obtained damping ratio and control are shown below.

	Obtained ξ	Control Gain
Modified IFF	0.83	1.99
IFF with k_p	0.83	2.02

5.4 Passive Damping - Critical Damping

$$\xi = \frac{c}{2\sqrt{km}} \tag{5.1}$$

Critical Damping corresponds to to $\xi = 1$, and thus:

$$c_{\rm crit} = 2\sqrt{km} \tag{5.2}$$

```
c_opt = 2*sqrt(k*m);
```

_____ Matlab ___

5.5 Transmissibility And Compliance

	_ Matlab
<pre>open('rotating_frame.slx');</pre>	

Matlab
%% Name of the Simulink File
<pre>mdl = 'rotating_frame';</pre>
%% Input/Output definition
clear io; io_i = 1;
io(io_i) = linio([mdl, '/dw'], 1, 'input'); io_i = io_i + 1;
io(io_i) = linio([mdl, '/fd'], 1, 'input'); io_i = io_i + 1;
<pre>io(io_i) = linio([mdl, '/Meas'], 1, 'output'); io_i = io_i + 1;</pre>

	Matlab
<pre>G_ol = linearize(mdl, io, 0);</pre>	
<pre>%% Input/Output definition G_ol.InputName = {'Dwx', 'Dwy', 'Fdx', 'Fdy'}; G_ol.OutputName = {'Dx', 'Dy'};</pre>	

5.5.1 Passive Damping

	Matlab
kp = 0;	
cp = 0;	
	Matlab
c_old = c;	
c = c_opt;	
G_pas = linearize(mdl, io, 0);	Matlab
%% Input/Output definition	
<pre>G_pas.InputName = {'Dwx', 'Dwy', 'Fdx', 'Fdy'};</pre>	
<pre>G_pas.OutputName = { 'Dx', 'Dy' };</pre>	

 $\overline{c = c_old};$

Matlab _

Matlab

Matlab

Matlab .

Matlab .

Kiff = opt_gain_iff/(wi + s)*tf(eye(2));

G_iff = linearize(mdl, io, ∅);

%% Input/Output definition
G_iff.InputName = {'Dwx', 'Dwy', 'Fdx', 'Fdy'};
G_iff.OutputName = {'Dx', 'Dy'};

kp = 5*m*W²; cp = 0.01;

Kiff = opt_gain_kp/s*tf(eye(2));

G_kp = linearize(mdl, io, 0);
%% Input/Output definition
G_kp.InputName = {'Dwx', 'Dwy', 'Fdx', 'Fdy'};
G_kp.OutputName = {'Dx', 'Dy'};



Figure 5.2: Comparison of the transmissibility



Figure 5.3: Comparison of the obtained Compliance

Notations

		Mathematical Nota	tion	M	atla	b	Unit
Actuator Stiffness		k		k			N/m
Actuator Damping	с					N/(m/s)	
Payload Mass	m					kg	
Damping Ratio $\xi = \frac{1}{2}$		$\xi = \frac{c}{2 \sqrt{hm}}$	= xi				
Actuator Force		F, F_u, F_v		F	Fu	Fv	Ν
Force Sensor signal		f, f_u, f_v		f	fu	fv	Ν
Relative Displacement		d, d_u, d_v		d	du	dv	m
Resonance freq. when $\Omega = 0$	ω_0 w			wØ		rad/s	
Rotation Speed	$\Omega = \dot{\theta}$	W				rad/s	
Low Pass Filter corner frequency ω_i				wi			rad/s
	Matl	hematical Notation	Matla	ab	U	nit	
Laplace variable	s		s				_
Complex number	j		j				
Frequency	ω		w		[ra	ad/s]	

Bibliography

- [1] T. Dehaeze and C. Collette. "Active Damping of Rotating Platforms using Integral Force Feedback". In: Proceedings of the International Conference on Modal Analysis Noise and Vibration Engineering (ISMA). 2020.
- [2] Thomas Dehaeze. Active Damping of Rotating Positioning Platforms. Source Code on Zonodo. July 2020. DOI: 10.5281/zenodo.3894342. URL: https://doi.org/10.5281/zenodo.3894342.
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